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ALCOA RESEARCH LABORATORIES

ALUMINUM COMPANY OF AMERICA

MECHANICAL PROPERTIES, INCLUDING FRACTURE TOUGHNESS AND FATIGUE, AND RESISTANCE TO STRESS-CORROSION CRACKING OF STRESS-FELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS

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ABSTRACT

The tensile and some compressive, shear and bearing properties of a total of 90 samples of 2014, 2024, 6061, 7075 and 7178 aluminum alloy extrusions in the TX51X temper have been determined. The extrusions ranged in thicknesses from 0.051 in. to 6.500 in. Ratios among these properties have also been computed.

Stress-corrosion tests of 18 samples of TX51X extrusions have been initiated.

TABLE OF CONTENTS

														Page
ı, ı I.	Introduction	•	•	•	•	y .	•	•	•	• .	•	•	•	1
II.	Material	•	•	٠	•	•	•	•	•	•	•	•	•	1
III.	Procedure	•	•	•	•	•	•	•	•	•		•	•	2
IV.	Summary	•		• ·	•	•	•	•	•	•	•	•	•	3
. V .	Tables and Fig	zui	es	3				•		•			•	5

THIRD QUARTERLY REPORT

MECHANICAL PROPERTIES, INCLUDING FRACTURE TOUGHNESS AND FATIGUE, AND RESISTANCE TO STRESS-CORROSION CRACKING OF STRESS-RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS

I. Introduction.

The tests being made under this contract are for use in establishing design mechanical properties. in MJL-HDBK-5A, including stress-strain and tangent-modulus curves, for 2014, 2024, 6061, 7075, 7079 and 7178 aluminum alloy extrusions in the TX51X tempers. For comparison, limited similar tests are being made of extrusions in the "heat-treated-by-user" temper. Also, some fracture-toughness, axial-stress fatigue and stress-corrosion tests are being made.

This Third Quarterly Report summarizes the results of tensile and some compressive, shear, bearing and stress-corrosion tests made to date on 90 samples of the various alloys in the TX51X temper. The samples ranged in thickness from 0.051 to 6.500 in.

II. Material.

A total of 92 samples of commercially-produced extrusions in the TX51X temper and 14 samples in the 0 temper have been received from two producers. The section thickness and identification of each sample is shown in Table I. Twelve of the samples in the 0 temper have been heat treated or heat treated and aged in accordance with applicable conditions in Mil-H-6088D.

III. Procedure.

Mechanical Properties

Tensile, compressive, shear and bearing specimens were taken at locations described in the Second Quarterly Report, dated September 15, 1966. The general dimensions of these specimens were shown in Figs. 1 to 3 inclusive of the Second Quarterly Report; the testing procedures are as outlined in the First Quarterly Report, dated June 15, 1966.

Some fracture toughness, axial-fatigue and tensile and compressive modulus specimens have been machined. Procedures for testing the fracture toughness and fatigue specimens are described in the First Quarterly Report.

The tensile and compressive specimens to be used for the modulus and stress-strain tests are as shown in Figs. 1 and 2. In all tests of longitudinal tensile specimens, strains will be measured over a 6-in. gage length with an Amsler-Martens mirror-type extensometer. In most of the tests of transverse tensile specimens it will be necessary to use smaller specimens and measure strains over a 4-, 2- or 1-in. gage length. The Amsler-Martens mirror-type extensometer will be used to measure strains over 4- and 2-in. gage lengths; the Tuckerman optical strain gage will be used for the 1-in. gage length. In all compressive tests, the Tuckerman optical strain gage will be used (2- or 1-in. gage length). The Amsler-Martens extensometer over the 6-in. gage length is probably ASTM Class A; over the 4-in. and 2-in. gage length it is an ASTM Class B-1. The Tuckerman extensometer is an ASTM Class A.

Resistance to Stress Corrosion

Stress-corrosion tests were initiated with specimens from 18 samples of TX51X extrusions. Additional samples were selected for corrosion testing, 6 in the TX51X temper and 7 in the heat-treated-by-user or heat-treated-and-aged-by-user temper. Test specimens were machined from these samples and these specimens will be stressed to the desired levels after determination of appropriate tensile properties. The specimens used and the procedures followed are as described in the First Quarterly Report.

IV. Summary.

The results of tensile and some compressive, shear and bearing tests of 90 samples of extrusions in the TX51X temper are as shown in Table II. The tensile properties of all samples exceed the applicable specified minimum values as shown in Table III.

The ratios among the tensile, compressive and shear properties are as shown in Table IV. The ratios among the bearing and tensile properties are as shown in Table V. The ratios among the properties at different locations with regard to width and thickness are as shown in Table VI. The ratios among bearing properties obtained using edgewise specimens to those using flatwise specimens, are shown in Table VII.

The current status of the stress-corrosion tests is given in Table VIII. Preliminary test results show that stress-corrosion cracking has occurred only among specimens from the

2024-T3510 and 7075-T6510 alloy extrusions. Although tests have thus far progressed for a maximum period of only 30 days, these results indicate typical performance of the various materials.

For the same alloy and temper, there sometimes is considerable scatter among certain stresses or ratios of certain stresses. In some cases, it is obvious that there is a trend with thickness. However, trends may also be dependent on other variables, but until a larger percentage of the total number of samples has been tested, it does not seem desirable to try to analyze this situation thoroughly.

D. J. Brownill

R. E. Davies

D. O. Sprowls

Tables and Figures.

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TABLE II Sheet 5 (Concluded)

TABLE III

THE PROPERTY OF THE PROPERTY O

SPECIFIED MINIMUM VALUES FOR ALUMINUM ALLOY EXTRUSIONS TESTED
[ÅF 33(615)-3580]

Federa:1 Specification	QQ-A-200/2b	QQ-A- 200/5b		QQ-A-200,8b	QQ-A-200/11b	Кле	00-A-200/13
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Alloy and Temper	2014-16510	2024-13510, -13511	-T8510, -T8511	6061-16510	7075-16510	-173510	7178-16510

* Offset equals 0.2 per cent.

MELE IV
ANCHO THE TRESILE, CONTRESSIVE AND SHEAR PROPERTIES
OF STRESS-PELIEVED STRETTEED ALLOUTEN ALLOY ELUKTSIONS
(AF7576(A5)-7590)

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TABLE IV

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APTOS ANCHO TER TERRITE, CONFRESSIVE AND SERAR PRIMERTOS OF STRESS-RELIEVED STRETCHED ALLOHOM ALLOY EXTRACTIONS ANTHON THE STRESS-RELIEVED STRETCHED TO THE STREET

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TABLE IV Sheet C (Continued)

TABLE IV (Concluded) RATICS ANCHE THE TENSIES, COMPRESSIVE AND FFRAR PROFERIES OF STRESS-RELIEVED STRENGED ALMOIDER ALL EXCHUSIONS (ARTS (615)-7580

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TABLE V (Concluded)

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1		246							1	Rearing specimen failed before reaching yield stress (2 per cent offset).	clasen fa	died bei	ore res	thing The	eld stre	ss (2 pe	r cent o	ffaet)	
	CELICAS		The property of the contract o						Motor	1000	tadinal	AL - ALL :	mg-Trans	gverse					
	10.6	T SIBURO T	FOR PRODUCE	X						1									

TABLE VI RALIOS AMONG THE MECHANICAL PROPERTIES AT DIFFERENT LOCALTORS |AP35(615)-3580|

						•						
	Samply	0								Bea	Bearing	
Alloy and Temper	Section Thickness,	Rumber	Direction*	Incation:	Tensile Ultimate Stress	Tensile Yield Stress	Compressive Yield Stress	Shear Ultimate Stress		Ultimate Stress e/D=1.5 e/D=2.0	71eld Stress e/5=1.5 e/f=2.(1755 e/_=2.(
2014-16510	1.657	318046	ı	F 100	1.01	1.01	τοτ	1.8	66.0	36.2	د.ع	c.97
2024-13510	6.525	318c2c**	ы	:≠I= ×	6.9	1.00		36.3	1	1	1	1
	2,130	318077	且日	- 310 310	9.3	1.32 0.96	1 3.	1.0.1	1 %	1.1	160	183
2C21-T3511	1.450	**15051£.	티니	। ह्या⊐ श्रीकः ।	1.0 1.09	0.9 8.9	0.99 0.99	۱ %	1 %	1 %	1 %	185
		*******	Ē	!	5	1.5	15	l	0.99	± <i>1</i> 6.0	€.50	\$56° 0
2024-T3510	2.520	313133**	il ia	1	30.1	3.0.	10:1	1 1	: :	1 1	1 1	1 1
	2.760	313c48	ដួរ	1 200	1.00	1.02	181	18	1.03	: 3:	10.1	1.5
			ង		. Б) Н	1.06	1.0A	10.1	\$ 8,8	0.97 0.99	1.02÷ 0.97	2007 2007 2007
8021-TB510	C.522	₹.,4805£₹	μÏ	.બાવ્ય (ઓપ્ય	6.9	96.0	6.9	86.	:		;	ł
	2,150	318078	£1.4	:elcv	પ્ ં રક્ષ	5.8 8	1.00	18	1 %	18	183	1.00
2004-4551	() 화 전	318025**	불교	No.	88	88	86.3 86.3	1.8	1 %	1.02	0.98	1.91
2024-13516	3,760	318079	텱ᆈ	- T- C-	1.1 2.1 2.1	1.02	1.00	118	6.99 1.05	1.01	1.01	1.01 9.98
•			Ħ		1.01	1.02	66.0	10:1	1.02	0.97	0.97	1.03
					_							

(Continued)

TABD: VI

RATIOS ANGRO THE RECHANICAL PROPERTIES AT INTEREST LOCALIONS
[APS7(615)-7580]

	•				,	1						
	Section	-			Tensile	Tensile	COMPRESSIVE	Sheer		Se .	Salasa	
Alloy and Temper	Thickness,	Number	Direction.	Location	Ultimate Stress	Yleld Stress	Yield	Ultimate Stress			2 511-5%	#1eld Stress
5061-16510	1.240	317907	н	N/A	1.01	1.01	96.0	6.9	1	1	ı	i
	1.960	317896	ы	,γ[ε: ι γ[ε: ι	1.01	1.01	3.02	96.0	6.9	1.00	8.5	0.0
	9-500	317897	н	E-40	0.99	86.0	8.0	8.0	86°0	c.38	6.9	2 89
			ដ		56°0	0.99	<i>16</i> °0	66.0	1	1	1	i
7375-16510	1.168	31786c	н	isto Sta	36.0	Ů,	76.0	\$.0	6.9	6. .3	8.3	% 3
•					; }	i	ł	1	1.02*	\$	1.02	1.024
•			Ħ		86.0	86.0	66.0	86.0	1.00	1.0	8. 8.	ر. ور
					!	1	}	ı	1.03#	1.0 4	# 8.0	9
	2.000	317861	H ·	E-11/	96.3	0.97	96.0	6.91	0.95	.9 86	86.	c.37
	261.6	318137**	H		0.97	96.0						
			ļ	.	1.03	66.0						
	3,040	315130		Elov	79.97	c.97						
			N. G	4.0	6.9	8.1						
7075-172510	2.000	317948		eden Edin	8.	96°0	86.0	6.98	1.01	86.0	1.02	8
2176-16510	0.625	756775	joa Joan	: #\c/s	86.3	ა 86.	86.0	1.00	96.0	1.01	66.0	96.0
	,		1 3.	2	9 G	86	56.0	1.8	1	ı	ı	:
	1.200	210139**	a fi	ijo da	8.8	8,8						
•	2,180	318140**	អ្នក	A PARTY	0.01	8.8						
					,		¥					

* L = Longitudinal) IR = Long-Transverse * Thickness; % = Midth * Cd-enise bearin - Spoinens; others = flatvise apecimens ** Froducer B; all others from Producer A

TABLE VII

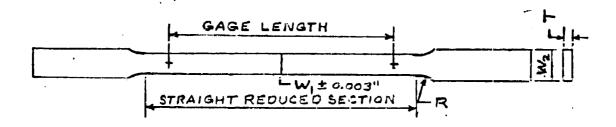
RATIOS OF BEARING PROPERTIES IN THE EDGEWISE DIRECTION TO THOSE IN THE FLATWISE DIRECTION FOR STRESS-RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS

	Section					Edgewise/	Flatvise	
Alloy and Temper	Thickness, in.	Number	Locetion*	Direction	BUS(E) e/D=1.5			(BYS(F) e/D=2.C
2024-T3510 2024-T3511	1.150	318077 317946 318021‡	1/2, W/4 1/2, W/4 1/2, W/4	러러다	0.97		20.00	• • •
2024- T3 510	2.760	318048	ν'∓ (Λ Σ΄Σ΄ Σ΄	거니티니	2100 2009	2400 2400	0000 0000 0000	3,000 3,000 3,000 3,000
2024-1851.0 2024-18511	1.200	318078 317895 318025‡	4/N,0/H 4/N,0/H 4/N,0/H	ннн			0 H 0 0	
2024_T8510	2.760	318079		ari 달다	0110 0000 0000	2000 2000	0 & 0 * 0 * 0 & 0 *	20011 200010 200010
6061-15510	1.240 1.960 6.500	317907 317896 317897	1/2, W/2 1/2, W/2 1/4	בהרר	0040 8888	00100 8888 9988	0.01.0 60.09	00000
7075 - 7510	1,188	317860	T/2,W/4 T/2,W/2	터니테니	98.48	0000 2888	0000 0000 0000	2000 2000 2000 2000

T. Thickness; W.- Width.
 L. Longitudinal; II - Long-Transverse
 Producer B; all others from producer A
 Bearing specimen failed before reaching yield stress (2 per cent offset).

RESISTANCE TO STRESS-CORROSION CRACKING OF STRESS-RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS

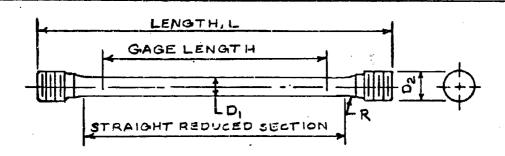
*	short Transverse F/N Days	2/2 6, 6	14 14 1 42	0/2 16	2/2 6, 6	1		port. osed. duplicate 0.050"	
anat!	75 -1	1 100			1 1 1 1 1	• •	•	Report exposed of dupl	
Stress 75% Yield Strenath*	Transverse Days	300	7	ତ୍ର କ୍ରନ୍ତିକ୍ର ଜୁନ୍ଦିନ୍ଦ୍ର	17,30 30 30	30	30	Quarterly or number by tests	
ss 75%	Long	2000	7 777	0000	0/2	0/2	0/2	he first ailed ove obtained <u></u>	
Stre	udinal Days	9999	ର ଚନ୍ଦ୍ର	ଚ ଚଚ୍ଚଚ	000	30	30	d in the mens fai ngths ob h, psi	000
	Longitudinal F/N+ Days	222	2 222	0 0000 0 0000	777	0/2	0/2	e described in r of specimens yield strengths specimens. Yield Strength, ps	65 650 67 650 67 650 67 650 600 600 600 600 600 600 600 600 600
ple	Number	317942 317926 317944+	317946++ 317890 317892 317893++	317895++ 317953 317927 317907	317954 317859 317860++	317900 317910	317997	ens used are denotes number of transverse yielder tension spectors.	944 946 893 895, 860
Section	Thickness, In.	0.255 0.510 0.950	1.200 0.255 0.510 0.950	0.315 0.375 1.240	0.375 0.438 1.188	0.375	0.625	Specimens used are F/N denotes number Short transverse y diameter tension specification 317944 317946 317893 317895, 317860	
	Alloy	2024-T3510	2024-T8510	6061-76510	7075-T6510	7075-173510	7178-T6510	Notes: * + + + + + + + + + + + + + + + + + +	



WIDTH	1, IN.	GAGE	REDUCED	RADIUS	THICKNESS
W	W ₂	LENGTH,	LENGTHIN	1,27.	IN.
0.500±0.003	3/4	6.000±0.002	7*	7/8	Z 0.499
0,250±0.002	3/8	1.000±0.002	1-1/2	3/8	20.250

* FOR SOME LONG-TRANSVERSE SPECIMENS, GAGE LENGTHS ARE 4IN., REDUCED-SECTION LENGTHS ARE 5 IN.

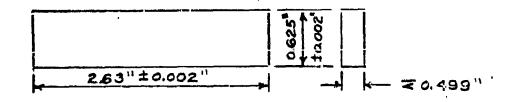
Sheet-Type Specimens



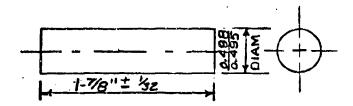
DIAMETER,	1N.	GAGE	REDUCED	RACIUS	LENGTH
בם	D2	LENGTH, IN.	LEMOTH, IN.	(R).	いだい
0.500±0.003	3/4	6.000 ± 0.002	7	5/8	9-1/2
0.500±0.003	3/4	4,000±0.002	5	5/8	7.1/2
0.500to.003	3/4	2.000±0.002	3	<i>5/8</i>	5-1/2
0.438 ±0.003	5/8	2.000±0.002	2-7/8	> D₁	5-1/4
0.375 £0.003	2/16	2.000±0.002	2-3/4	50,	. 5

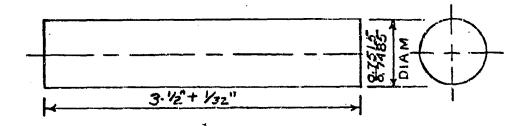
Round Specimens

Fig. 1 General Dimensions of Tensile Specimens For Modulus and Stress-Strain Tests



Sheet-Type Specimen





Round Specimens

Fig. 2 General Dimensions of Compressive Specimens
For Modulus and Stress-Strain Tests